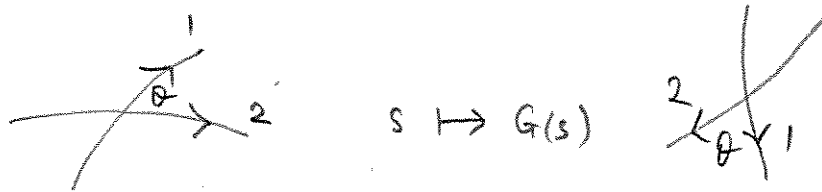


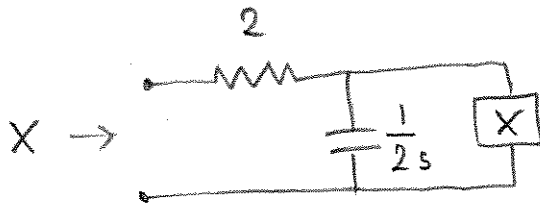
4 F1 solutions (2017)

- (a) A mapping $s \mapsto G(s)$ is conformal if $G(s)$ is analytic and $\frac{d}{ds} G(s) \neq 0$. A conformal mapping preserves angle and sense:



- Usefulness: (1) justification of rule for breakaway points in the root-locus diagram,
 (2) informal proof of Nyquist stability criterion,
 (3) estimate of location of dominant poles in a feedback system.

(b) (i)



Impedance of parallel connection = $(2s + X^{-1})^{-1}$

$$(ii) \quad 1 = (X - 2)(2s + X^{-1})$$

$$= 2sX + 1 - 4s - 2X^{-1}$$

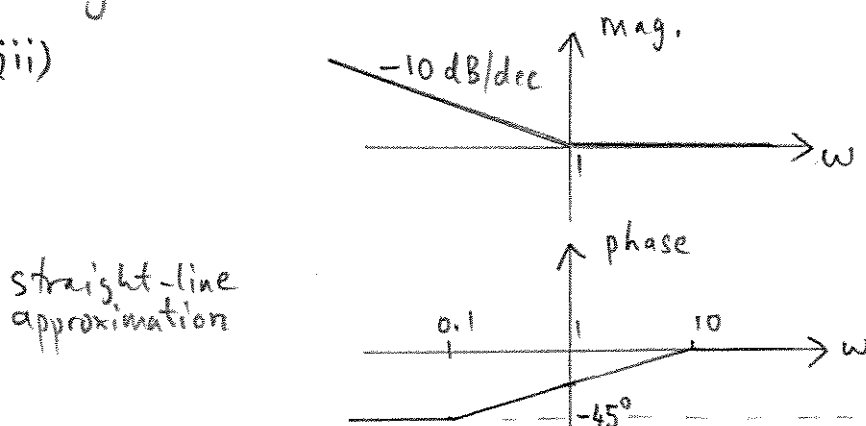
$$\Rightarrow 2sX^2 - 4sX - 2 = 0$$

$$\Rightarrow X = \frac{4s \pm \sqrt{16s^2 + 16s}}{4s} = 1 \pm \sqrt{1 + \frac{1}{s}}$$

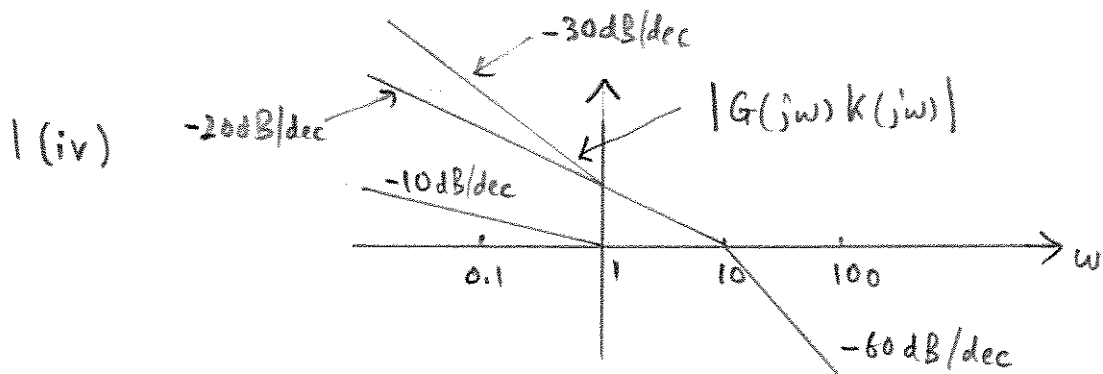
$$\Rightarrow Z = \pm \sqrt{\frac{s+1}{s}}$$

For large s circuit behaves like a 1Ω resistor, hence $+$ sign is the correct one.

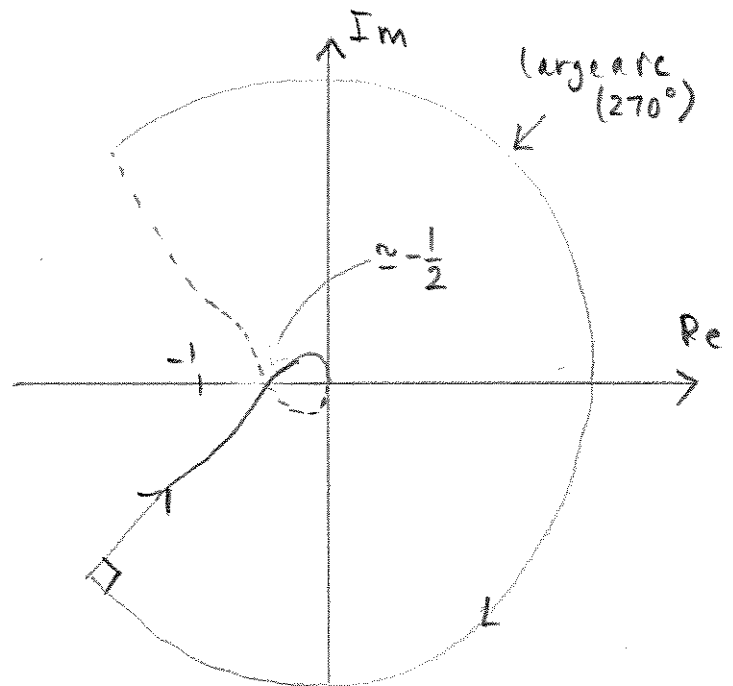
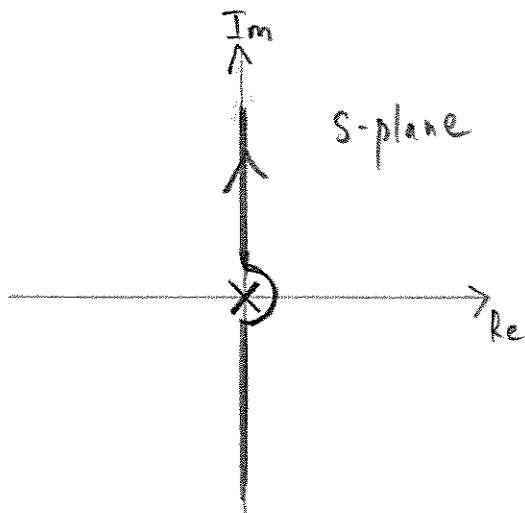
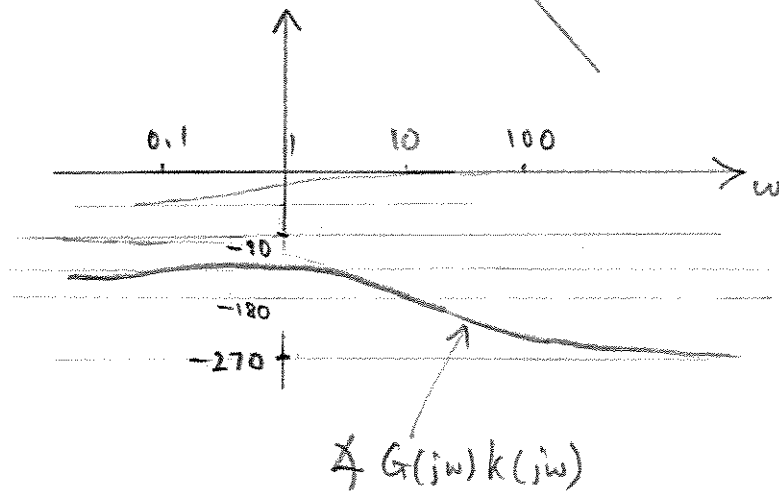
(iii)



$$Z(j\omega) = \sqrt{\frac{1+j\omega}{j\omega}}$$



Bode diagram

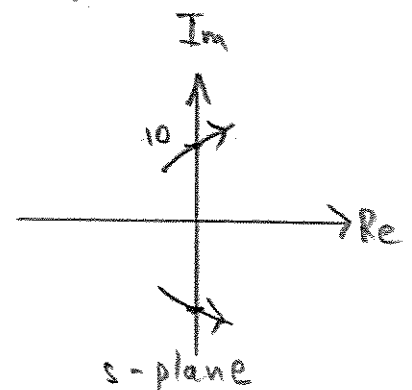


$$\angle G(j10)k(j10) \approx -180^\circ$$

$$|G(j10)k(j10)| \approx \frac{1}{2}$$

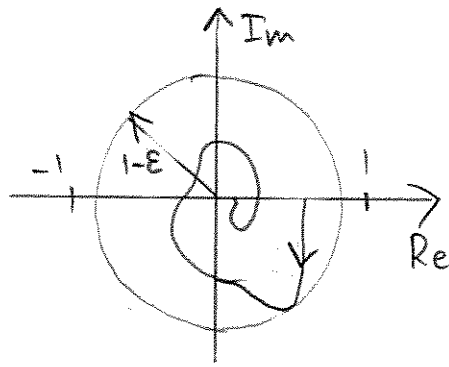
- (v) Closed-loop stable for
- | |
|------------------|
| $0 < k < 2$ |
| $2 < k < \infty$ |
| $k < 0$ |
- 2 RHP poles
- 1 RHP pole

Locus of solutions mapped by $G(s)K(s)$ to -0.5



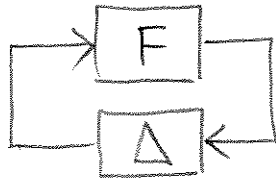
- (vi) A pair of dominant poles moves from the LHP to RHP with frequency increasing through 10 rad/sec as k increases through 2 .

2(a)(i)



Nyquist diagram of $F(s)\Delta(s)$ (stable) lies inside a circle of radius $1-\epsilon$, and hence cannot encircle ± 1 point. Feedback system is stable by Nyquist stability criterion.

(ii)



A closed-loop transfer function:

$$\frac{1}{1 - F\Delta}$$

Let $c = \frac{\pm 1}{|F(j\omega_0)|}$ and $-\pi < \angle \left[\frac{1 - j\omega_0 T}{1 + j\omega_0 T} \right] \leq 0$

for $T \gg 0$. Hence by choice of $\pm$ sign and $T \gg 0$ we can always ensure that

$$F(j\omega_0) \Delta(j\omega_0) = 1$$

which makes the loop unstable. Note that

$$|\Delta(j\omega)| = \frac{1}{|F(j\omega)|} = f(\omega_0) \leq f(\omega)$$

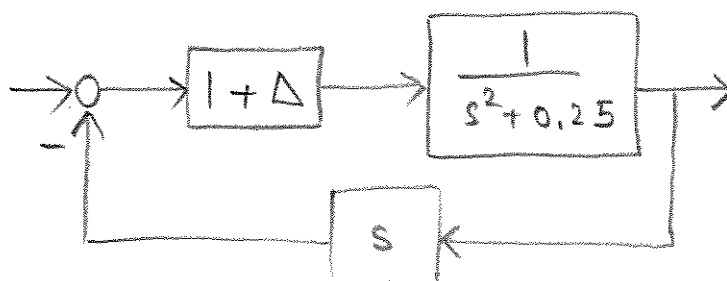
for all ω , so the condition of (a)(i) is satisfied.

(b) Blockwork - block diagram rearrangement.
Necessary and sufficient condition:

$$\left| \left(\frac{G_0 k}{1 + G_0 k} \right) (j\omega) \right| < \frac{1}{h(\omega)}$$

for all ω .

(c)(i)



(c)(ii)

$$\frac{G_0 k}{1 + G_0 k} = \frac{\frac{s}{s^2 + 0.25}}{1 + \frac{s}{s^2 + 0.25}} = \frac{s}{s^2 + s + 0.25}$$

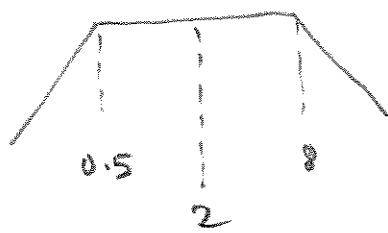
Need

$$\delta_0 \left| \left(\frac{j\omega + 0.5}{j\omega + 0.8} \right) \left(\frac{j\omega}{(j\omega + 0.5)^2} \right) \right| \leq 1 \quad (1)$$

for all ω .

$$(1) \Leftrightarrow \delta_0 \frac{\omega}{|(j\omega + 0.5)(j\omega + 8)|} \leq 1$$

Bode magnitude:



$\uparrow$
geometric mean $\rightarrow$ maximum

$$\text{Hence, (1)} \Leftrightarrow \delta_0 \frac{2}{|(j2 + 0.5)(j2 + 8)|} \leq 1$$

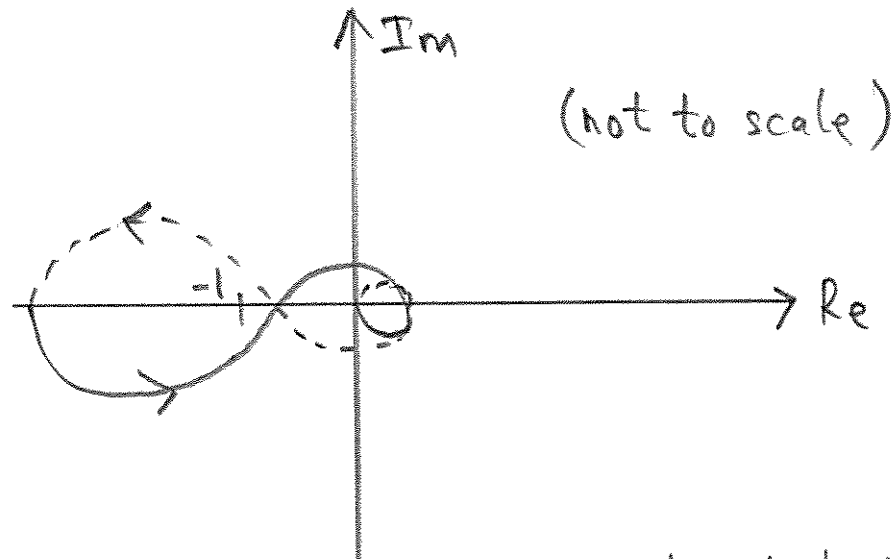
$$\Leftrightarrow \frac{2\delta_0}{0.5(1+4^2)2} = \frac{2\delta_0}{17} \leq 1$$

$$\Leftrightarrow \delta_0 \leq 8.5$$

Choose $\delta_0 = 8.5$.

(iii) Although (b) is a necessary and sufficient condition the form of $\Delta(s)$ in (c) is restricted, so result might be conservative.

3(a)(i)



(ii) Nyquist diagram shows one anti-clockwise encirclement of -1 point. Since feedback loop is stable with $k=1$ there must be one pole of $G(s)$ in the RHP.

(b)(ii) Phase excess rises (due to RHP pole) and then decreases to -180° indicating two RHP zeros.
Estimated location:

$$\begin{array}{ll} s = 10 & \text{RHP pole} \\ s = 200 & \text{2 RHP zeros} \end{array}$$

[Actual transfer function

$$G(s) = \frac{10(s-100-100j)(s-100+100j)}{(s+60)(s+100)(s-10)} \quad]$$

(ii) Crossover frequency couldn't be much lower than 10 rad/sec or much higher than 200 rad/sec.

$$\begin{aligned} (c)(i) \quad 20 \log_{10} 0.72 &= -2.85 \text{ dB} \\ 20 \log_{10} 1.8 &= 5.11 \text{ dB} \\ 2 \tan^{-1} 1.8 - 90 &= 31.89^\circ \end{aligned}$$

Gain reduced by 2.85 dB at $\omega = 20$ rad/sec so $\omega = 20$ is new gain crossover frequency.

(C)(i) cont.

Gain reduced by around 8 dB at $s=0$.

Gain increased by around 2 dB at high frequency.

Phase at $\omega=20$: $180 + 20 + 32$

$$\Rightarrow \text{phase margin} = 52^\circ$$

(ii) Gain at $s=0 \simeq 11 - 8 = 3 \text{ dB}$.

So gain reduction of 3 dB gives instability.

Phase = 180° at $\omega \simeq 50 \text{ rad/sec}$, where gain $\simeq -10 \text{ dB}$. So gain increase of 10 dB gives instability.

$$\Rightarrow \text{gain margin} = 3 \text{ dB}$$

(iii) Try a lag compensator

$$k_2(s) = \frac{s+8}{s+5.5}$$

Increases gain at $\omega=0$ by 3.25 dB.

Introduces phase lag at $\omega=20$ equal to:

$$\tan^{-1} \frac{20}{8} - \tan^{-1} \frac{20}{5.5} = -6.42^\circ$$

which should preserve phase margin $> 45^\circ$

Gain should be $> 6 \text{ dB}$ at $\omega=0$ and $< -10 \text{ dB}$ at $\omega=50$ so gain margin should be $> 6 \text{ dB}$.

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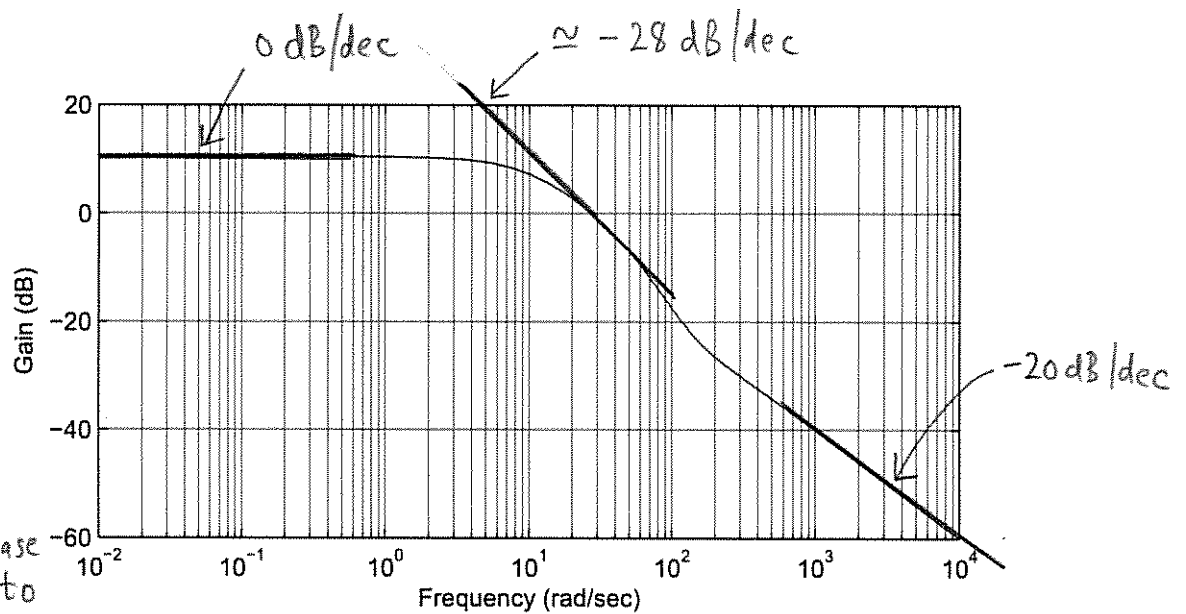
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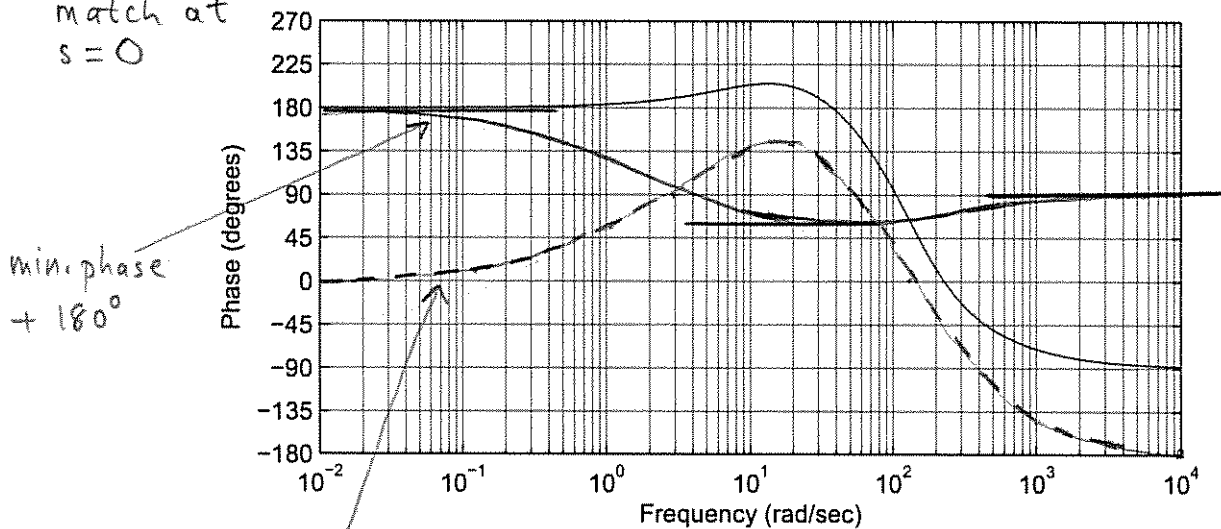
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? May 2017, Module 4F1, Question 3.

(b)(i)



add 180°
to min. phase
(-1 sign) to
match at
 $s=0$



Extra copy of Fig. 2: Bode diagram for Question 3.

Version MCS/1

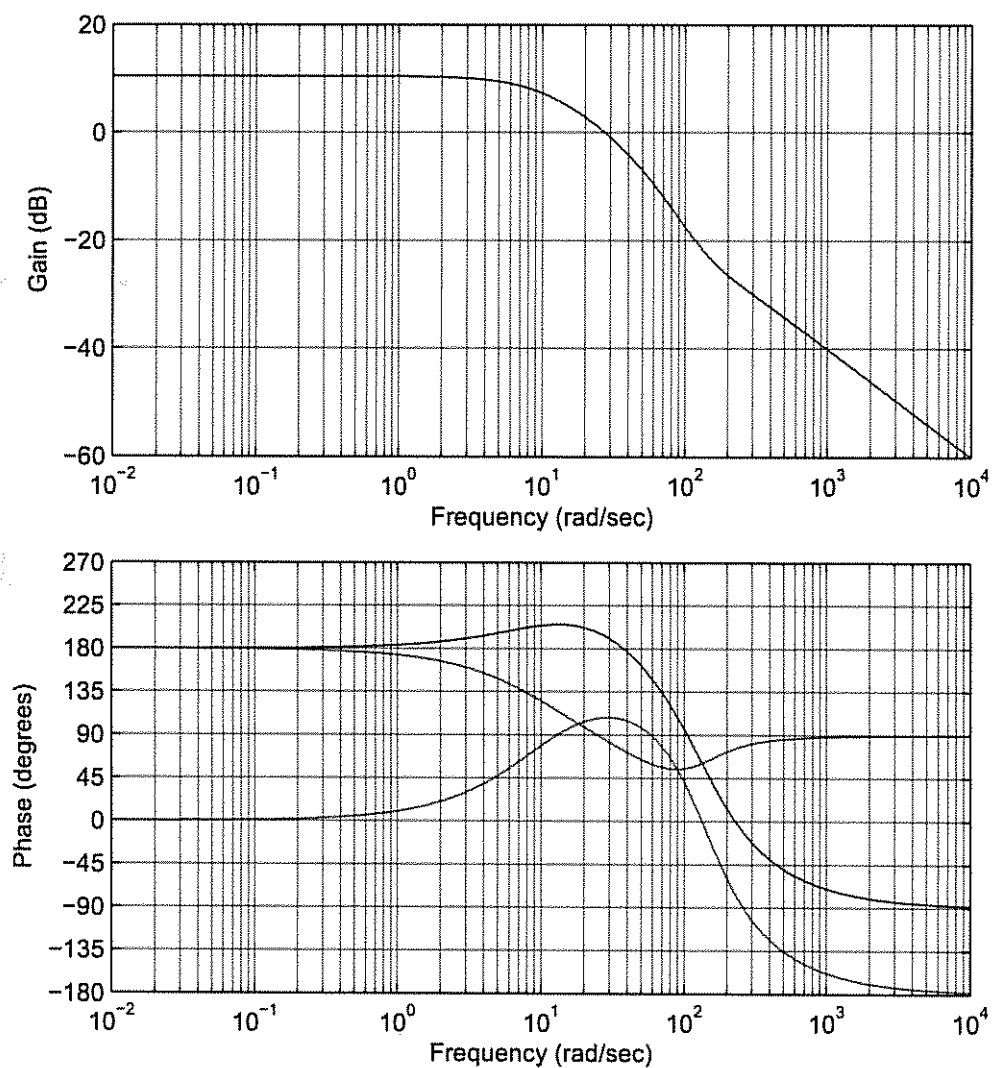
Candidate Number:

EGT3

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? May 2017, Module 4F1, Question 3.

(b)(i) Accurate computer plot.



Extra copy of Fig. 2: Bode diagram for Question 3.

Version MCS/1

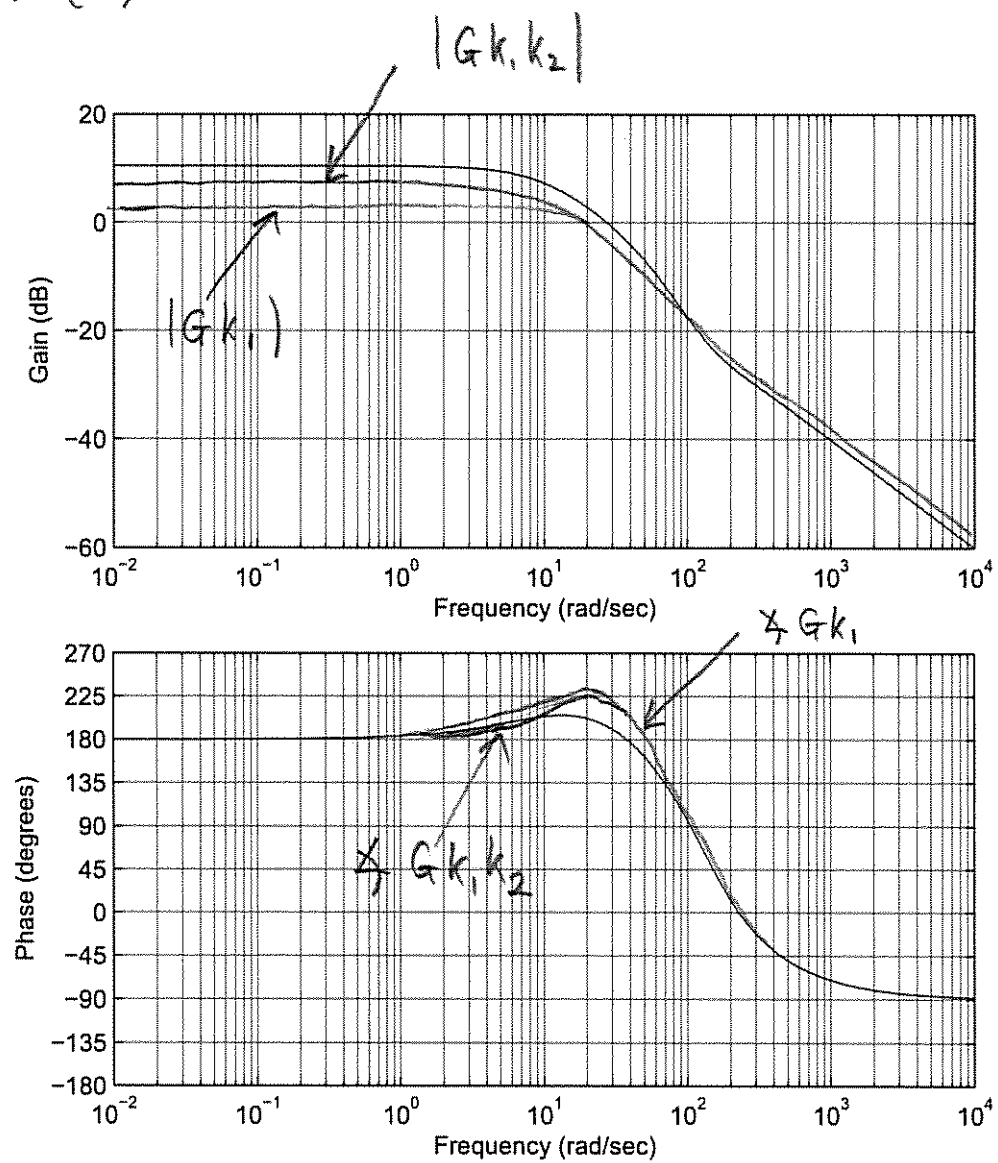
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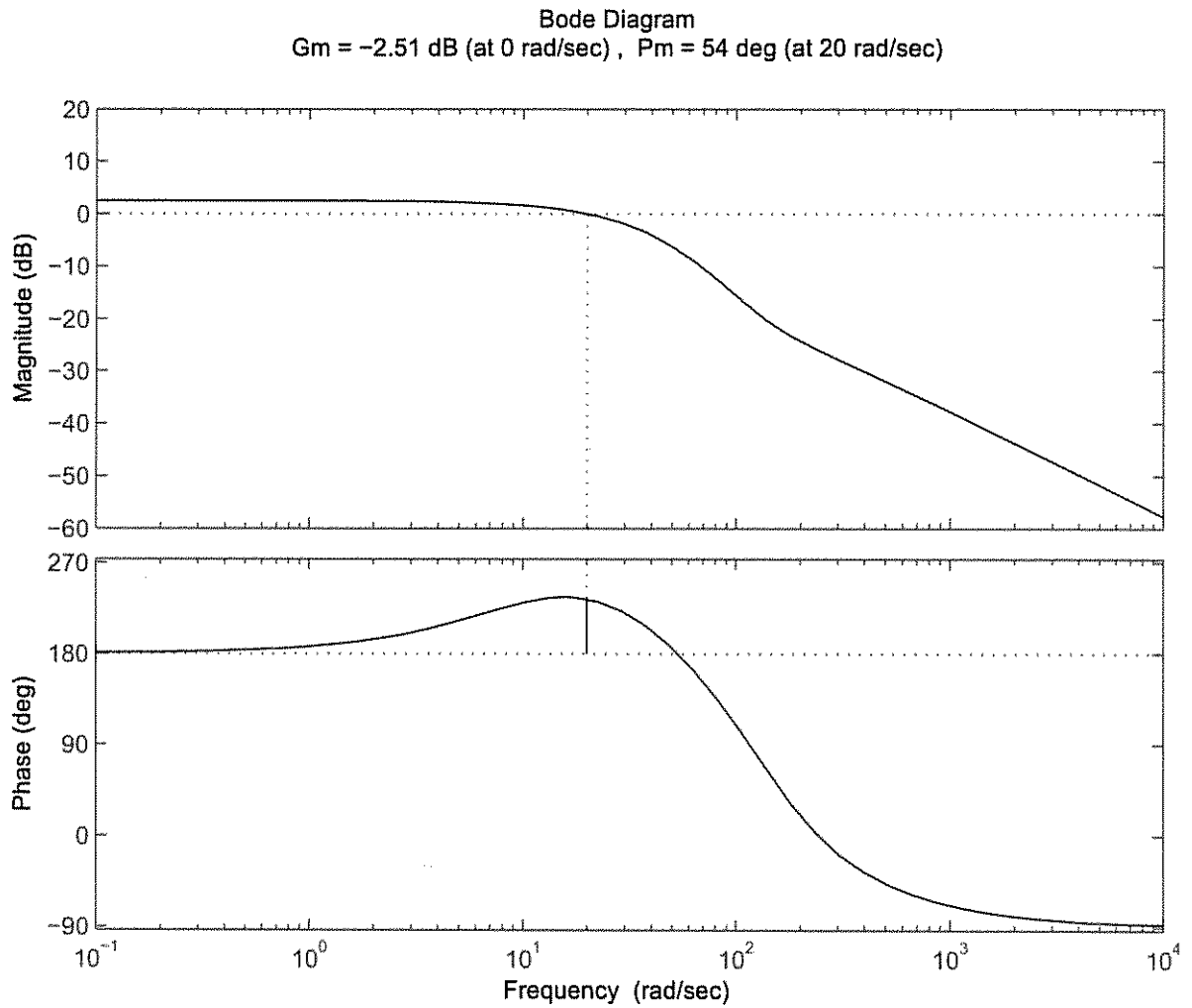
(c)(i) - (iii)



Extra copy of Fig. 2: Bode diagram for Question 3.

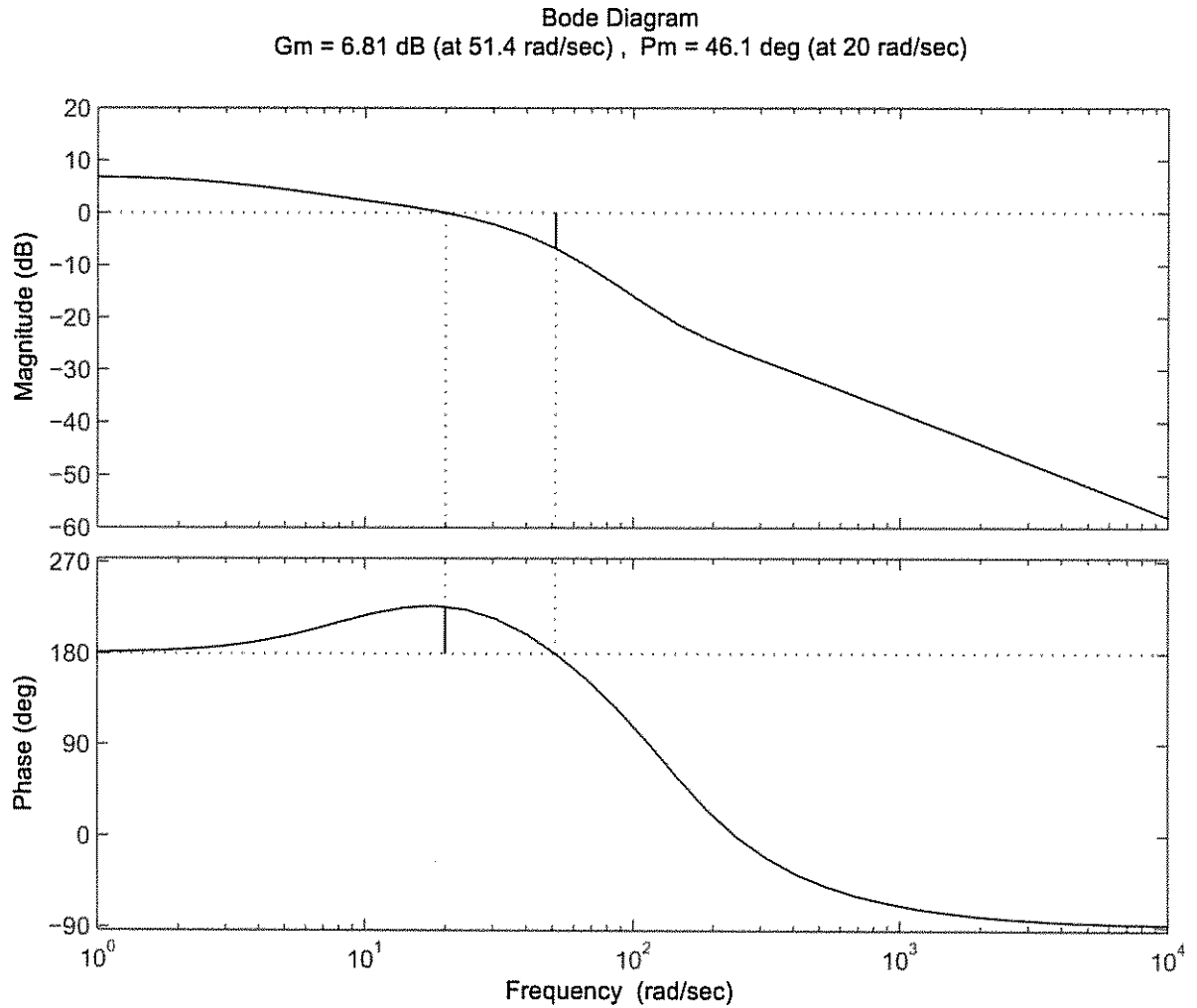
©(i) Accurate computer plot.

Version MCS/1



(c)(iii) Accurate computer plot.

Version MCS/1



$$\text{Actual } k_2(s) = 0.963 \frac{s+7}{s+4}$$

Version MCS/1

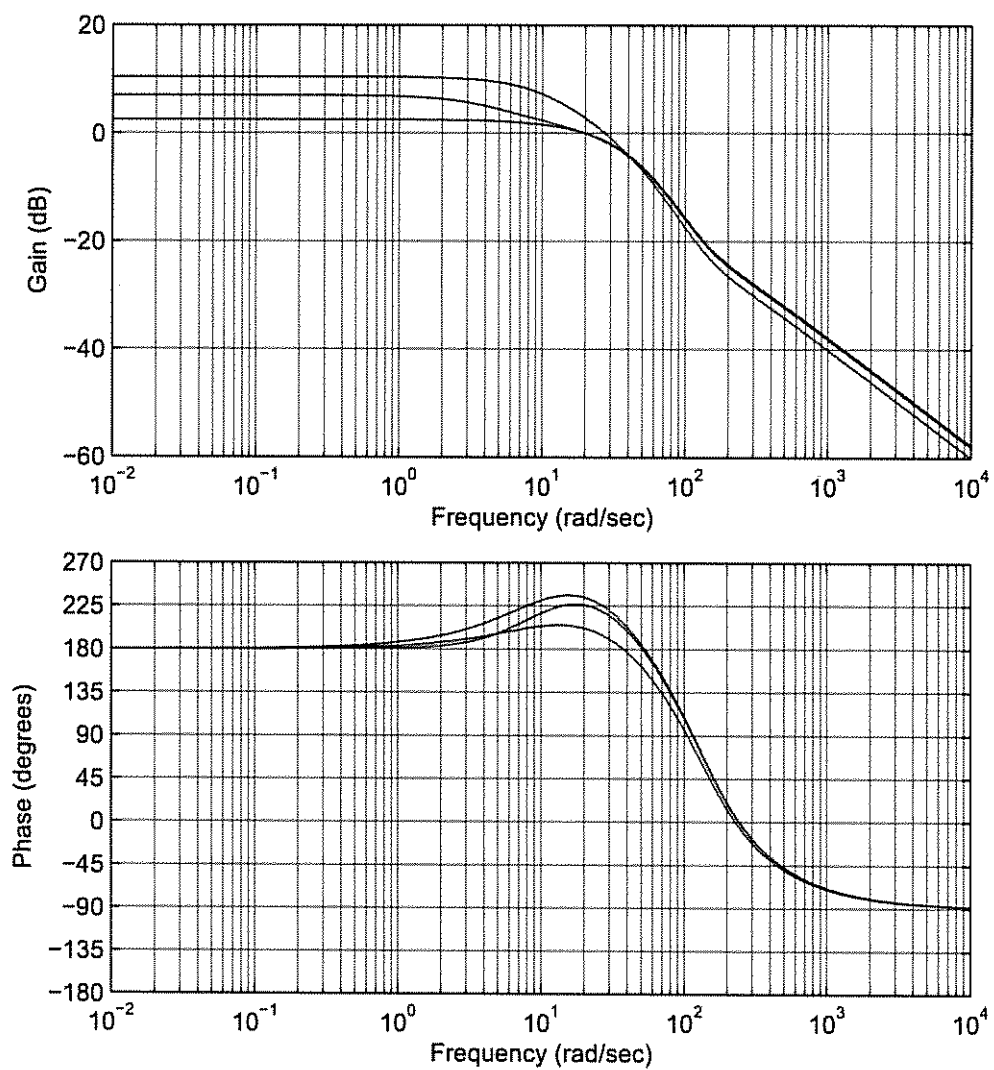
Candidate Number:

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(c)(i)-(iii) Accurate computer plot



Extra copy of Fig. 2: Bode diagram for Question 3.

ENGINEERING TRIPOS PART IIB 2017
ASSESSOR'S REPORT, MODULE 4F1
Control System Design

Q1. Conformal mapping. Non-rational compensator.

10 attempts, mean 12.3/20 (61.5%), maximum 19, minimum 5.

An unpopular question. Candidates may have been put off by a compensator which was a square-root of a rational function. Apart from a couple of attempts which didn't get far, most candidates made good progress and were successful to apply the standard principles to a new situation.

Q2. Robustness and VTOL aircraft.

27 attempts, mean 11.63/20 (58.15%), maximum 18, minimum 5.

A popular question but with rather mixed performance from candidates. The bookwork was somewhat poorly done, especially part 2(a)(ii). A lot of candidates got bogged down with lengthy calculations on part 2(c)(ii) and failed to see that a simple Bode plot would readily lead to a solution.

Q3. Bode gain/phase and compensator design.

27 attempts, mean 14/20 (70%), maximum 19, minimum 9.

A popular question with many first class answers that showed a very good grasp of this material.